Martian Ionized Carbon Dioxide and Atomic Oxygen Electrons Escaping the Atmosphere

Rudy A. Frahm

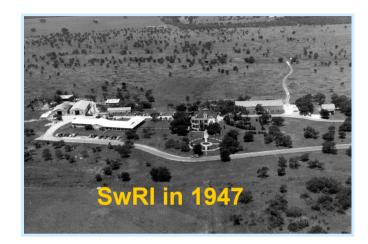
Southwest Research Institute

http://www.aspera-3.org rfrahm@swri.edu

Fu-Jen University, Taiwan

December 14, 2011

Introduction to Southwest Research Institute

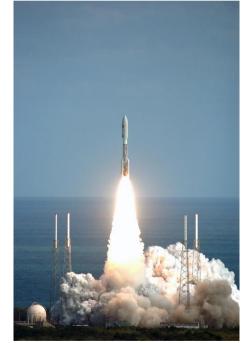




- Founded in 1947 as an Independent, Not for Profit, Applied Engineering and Physical Sciences Research and Development Institution
- Occupies a 1200 Acre
 Campus in Northwest
 San Antonio, Texas, USA
- Employs 3 000

Space Science & Engineering Division

- We conduct scientific research in space - <u>magnetospheric</u> and <u>heliospheric physics</u> as well as <u>planetary</u> and <u>solar astronomy</u>.
 - We design and build instruments for space research.
 - We sometimes build the entire payload.
 - We have numerous instruments on Earth-orbiting satellites and probes to other planets.
 - We also build rocket payloads.
 - We analyze the data.

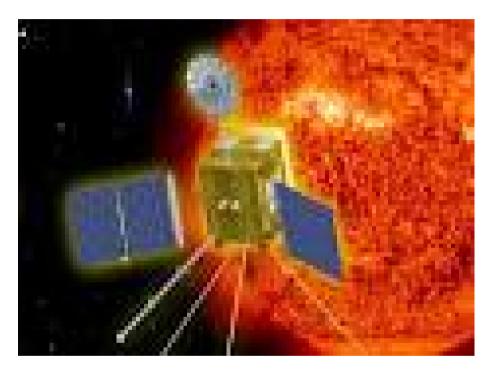


New Horizons departs for Pluto, 19 January 2006

Missions to the Sun

<u>ESA - Solar Orbiter</u> High Altitude (48 Solar Radii), Close-up Solar Observation

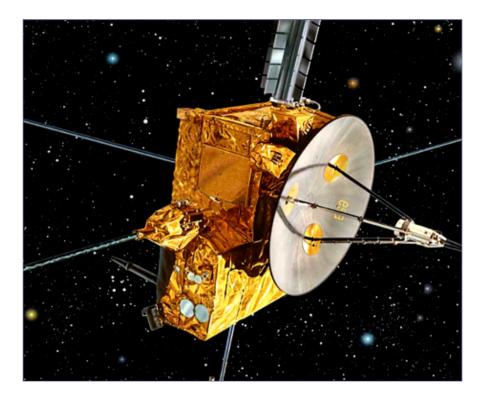
EUV Spectrometer PI: Don Hassler Launch in 2017 from Cape Canaveral, FL Launch Vehicle: Atlas V Mission Duration: 7 years



Solar Wind Missions

NASA - Solar Wind Plasma: Ulysses/ACE First observations of 3-D solar wind (Ulysses) and real-time advanced warning (ACE)

PI: Dave McComas Launched in 1990 and 1997 Are both still producing data.

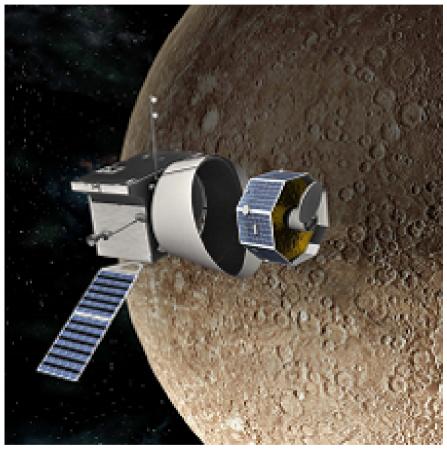


Missions to Mercury

ESA - Bepe Columbo: Mercury Polar Orbiter

Determine the origin and evolution of Mercury Launch: 2014 (6 year curse) Launch Vehicle: Soyuz-2B/Fregat M from French Guiana

STROFIO PI: Stefano Livi Neutral Particle Analyzer to determine remaining 70 % of observed atmospheric pressure

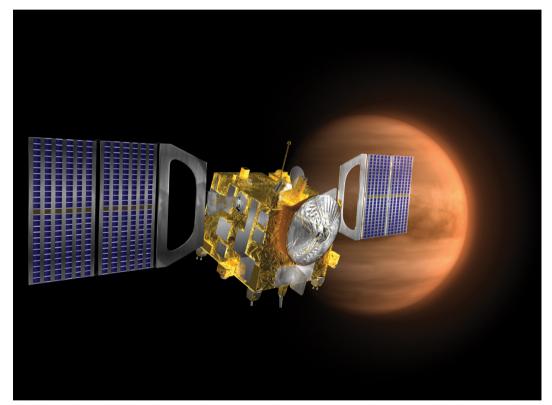


Missions to Venus

ESA – Venus Express

Venus complex dynamics and chemistry, interactions between the atmosphere and surface, and interactions between atmosphere and solar wind

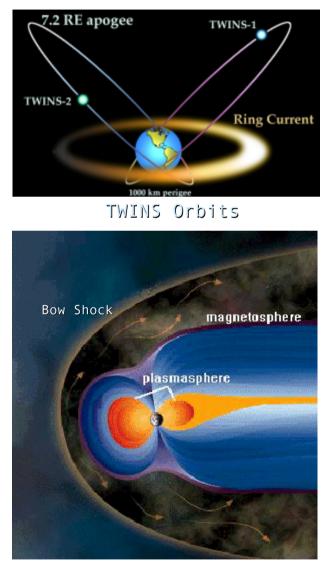
ELS/ASPERA-4 PI: J. David Winningham Launch: November 9, 2005 Launch Vehicle: Soyuz-Fregat Baikonur, Kazakhstan



Missions to Earth

DOE - TWINS First stereo imaging of the Earth's magnetosphere PI: Dave McComas Launches in 2008 (Mission of Opportunity on 2 DOE spacecraft)

NASA - Magnetospheric Multiscale PI: Jim Burch 4-spacecraft mission. Objective is to understand how particles are accelerated in the Magnetosphere. Launch in 2013.

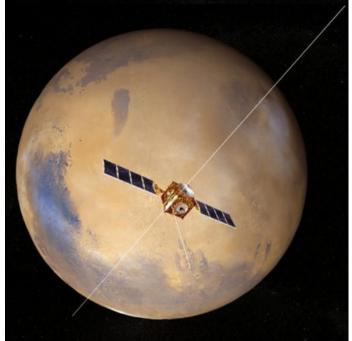


Earth's Magnetosphere

Missions to Mars

ESA – Mars Express Search for Mars Water Escape

ELS/ASPERA-3 PI: J. David Winningham Launch: June 2, 2003 Launch Vehicle: Soyuz-Fregat Baikonur, Kazakhstan





NASA – Mars Science Laboratory Search for organic life

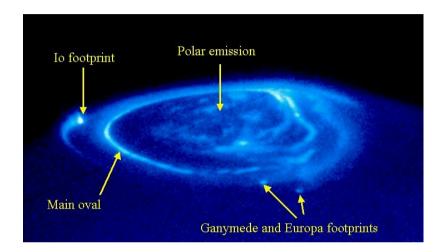
RAD – radiation detector PI: Don Hassler Launch: November 26, 2011 Launch Vehicle: Atlas V Cape Canaveral, FL

Missions to Jupiter

NASA - JUNO

Jupiter Polar Orbiter Mission PI: Scott Bolton Launch: August 5, 2011

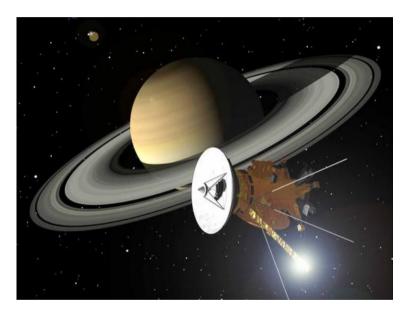
Juno investigates: the possible existence of an ice-rock core; the water abundance and deep wind profiles in the atmosphere; the origin of the magnetic field, and the dynamics of the polar magnetosphere and aurora.



Missions to Saturn

NASA - Cassini Plasma Spectrometer PI: Dave Young Ion-Neutral Mass Spectrometer Team Leader: Hunter Waite

Launched in 1997 In Saturn orbit since 2004 Prime mission through 2008.



Missions to Comets

ESA - Rosetta Comet Orbiter/Lander PI: ALICE (Alan Stern); IES (Jim Burch)

Launched in 2004, Arrives 2014 Comet: 67P Churyumov-Gerasimenko End of mission 2017



Missions to Pluto and Kuiper Belt

<u>NASA - New Horizons</u> First mission to Pluto PI: Alan Stern Launched January 2006 Arrival at Pluto in July 2015

New Horizons is the first mission to the last planet—the initial reconnaissance of Pluto-Charon and the Kuiper Belt, exploring the mysterious worlds at the edge of our solar system.

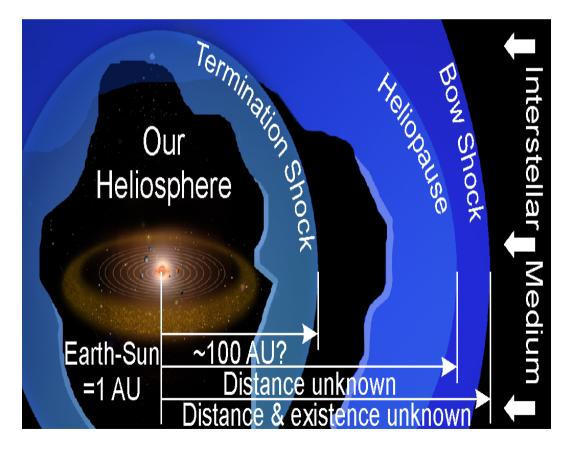


Intersteller Boundary Missions

NASA - Interstellar Boundary Explorer - IBEX

PI: Dave McComas Launch in 2008

IBEX globally images particles from the termination shock to determine the solar system's interaction with the galaxy.



Avionics Projects

Avionics for numerous science & defense missions

"Deep Impact" Mission to Comet Tembel

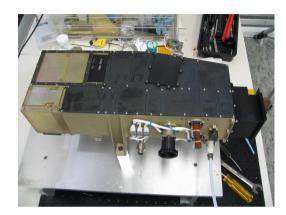


SwRI provided all of the Control and Data Handling electronics.RIU = Remote Interface UnitPICU = Primary Interface Computer Unit

Recently Developed Instruments



Rosetta/IES



New Horizons/Alice



Mars Express/ELS



New Horizons/Ralph (joint w/Ball Aerospace)



IMAGE/MENA



New Horizons/SWAP

SwRI Constructed Science Deck

NASA - IMAGE

First mission to image the Earth's magnetosphere. PI: Jim Burch Launched in 2000. In extended mission until 2010. Operation interrupted in Dec. 05.

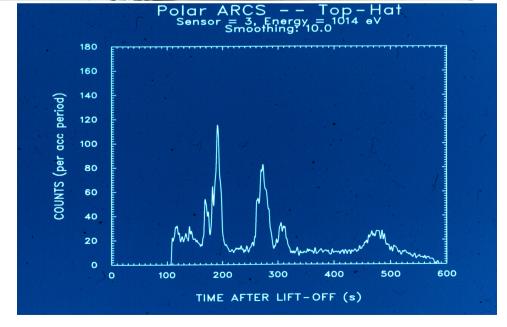




Rocketry







Young Engineers and Scientists - YES

Reaches high school age students and teachers in the San Antonio area.

It is a 3-week summer workshop here at SwRI.

Consists of research/engineering projects, electronics, basic computer skills, Power Point presentations, etc.

In some cases students pursue independent study with SwRI mentors during the school year.

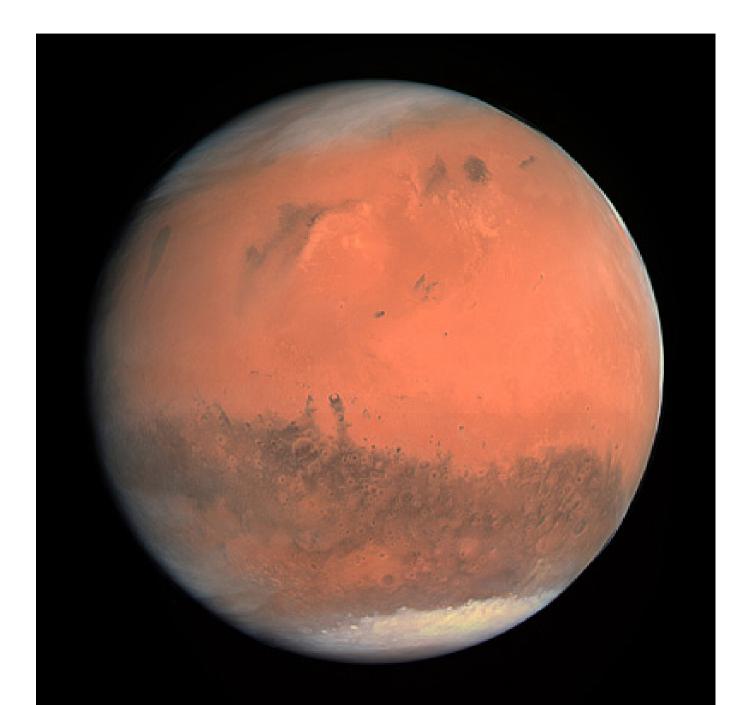
The goals of YES are to increase the number of students -- especially minorities and women -- seeking Science or Engineering careers, assist their choice of college and college major, and involve high school teachers in science and engineering research and technology that can be used in their classrooms.

Check out the web site: http://yesserver.space.swri.edu/

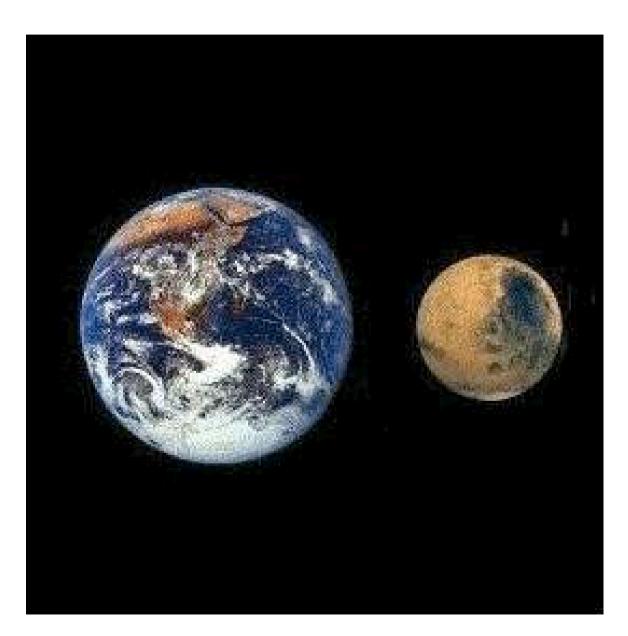
UTSA - SwRI Graduate Program Physics and Astronomy

Ten members of the science staff at SwRI are joint faculty members in the Department of Physics and Astronomy of the University of Texas at San Antonio.

Mars - The Red Planet



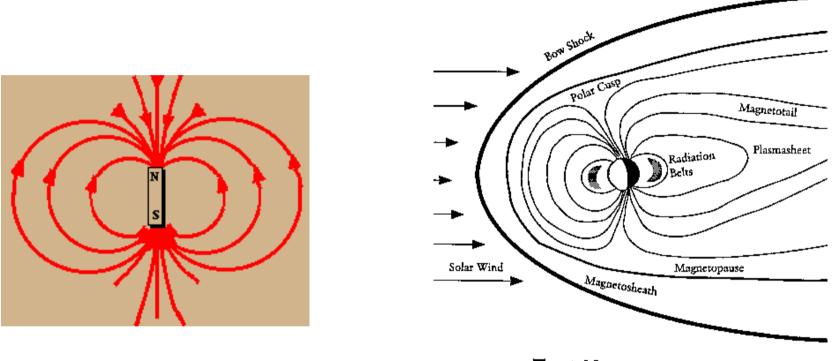
The Size of Mars Versus Earth



Mars

Earth

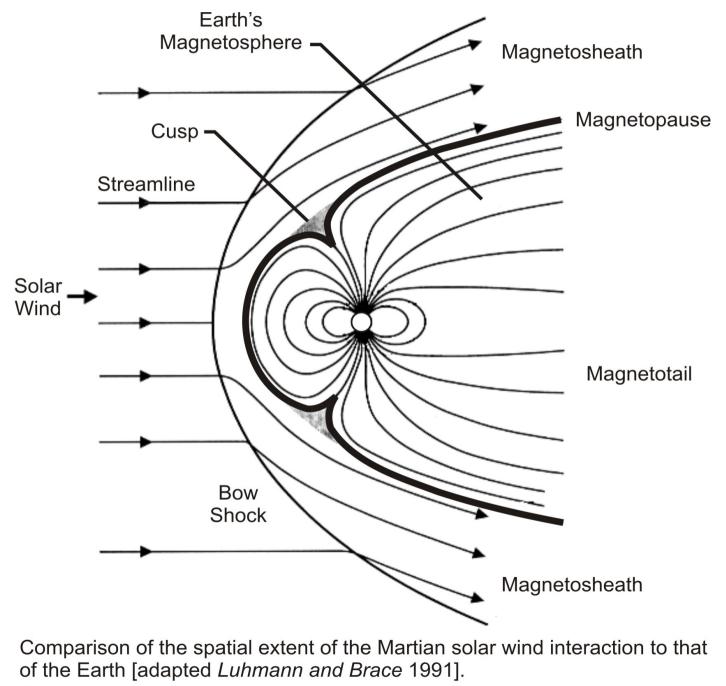
What is the <u>Magnetosphere</u>?



THE MAGNETOSPHERE

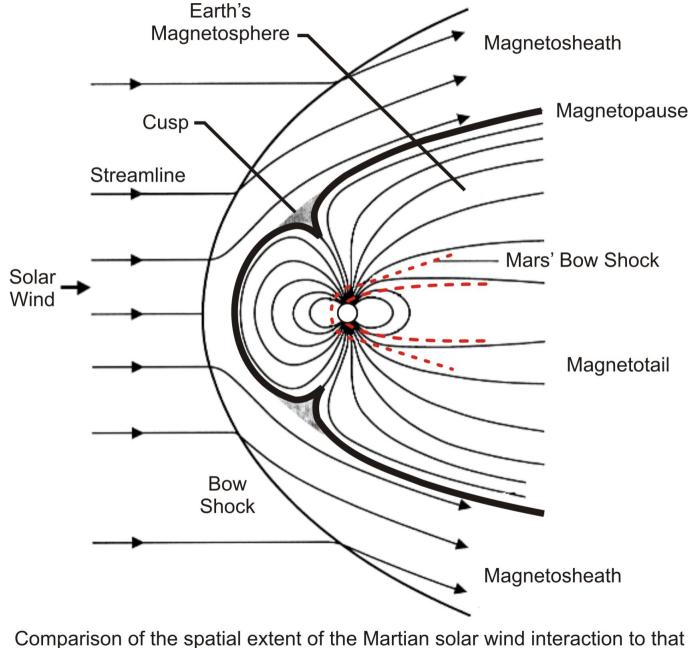
The presence of the solar wind re-shapes the magnetic field lines.

Magetosphere of the Earth



TA004550B

Magnetosphere of Mars (red dashed magnetic field lines)

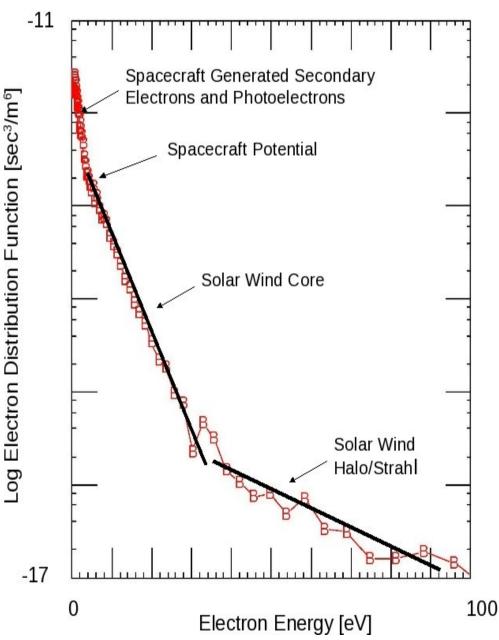


Comparison of the spatial extent of the Martian solar wind interaction to tha of the Earth [adapted *Luhmann and Brace* 1991].

Background on the Solar Wind

Solar Wind Core Electrons

- Generally, <50 eV
- Isotropic
- Solar Wind Halo Electrons
 - Generally, >50 eV
 - Isotropic
- Solar Wind Strahl Electrons
 - Generally, >50 eV
 - Aligned with the Magnetic Field
- Electron Distribution Accelerated by the Spacecraft Potential
- Spacecraft Generated Secondary Electrons and Photoelectrons are Observed Below the Spacecraft Potential



Penetration of Obstacle by the Solar Wind

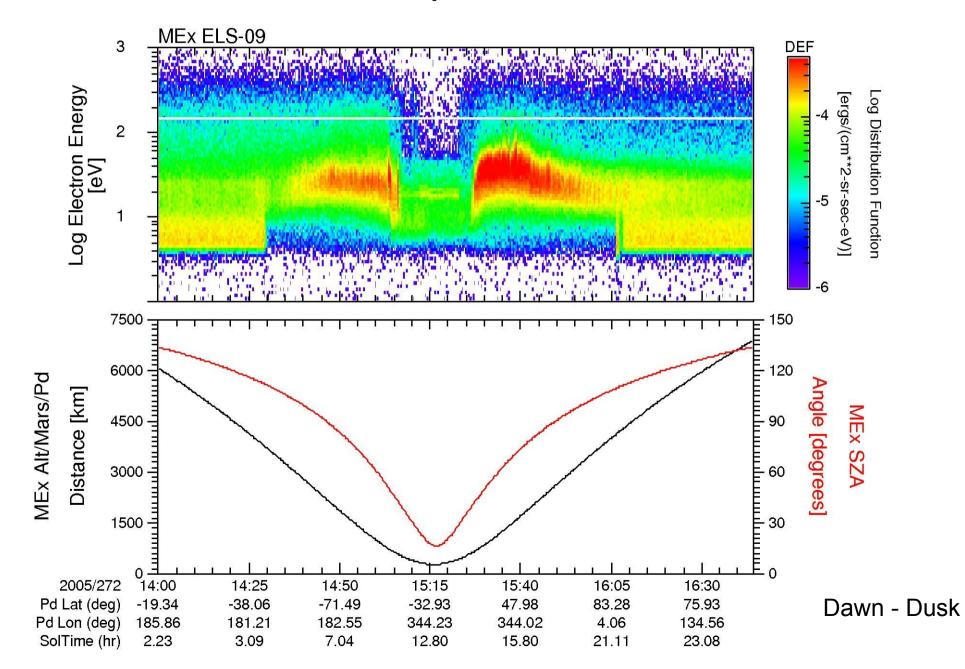
Solar Wind Penetrates to the Ionosphere of the Obstacle (Mars and Venus)

Solar Magnetic Field Distorts and is Draped Around the Obstacle

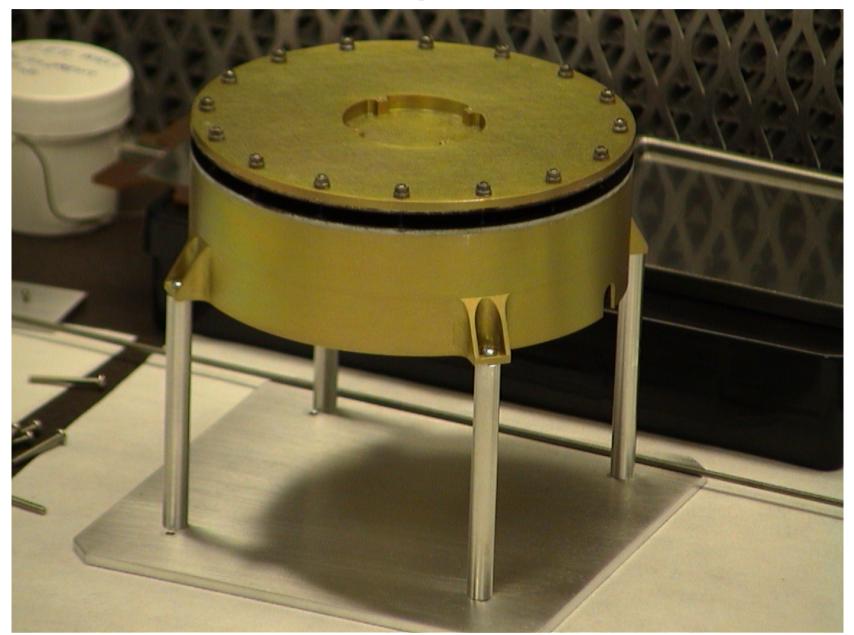
Solar Wind Core Influenced by the Shock Potential

Since the Bulk of the Solar Wind Halo/Strahl Population has Energies Above the Shock Potential, the Solar Wind Halo/Strahl is Observed to Penetrate (Mostly Undisturbed) Through the Shock, to the Ionosphere of the Planet

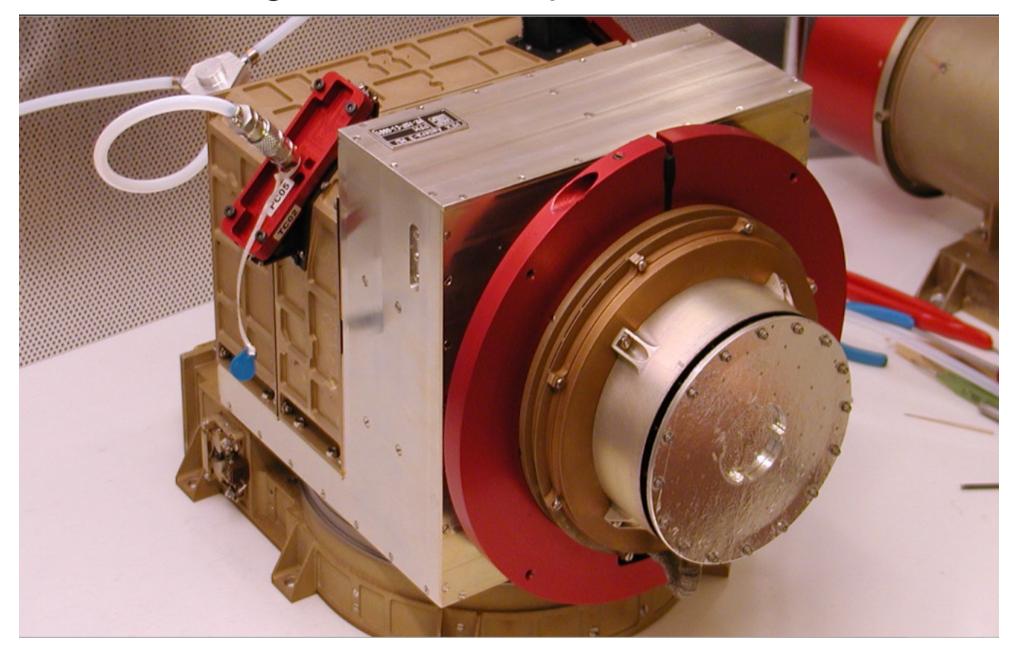
Mars Express View



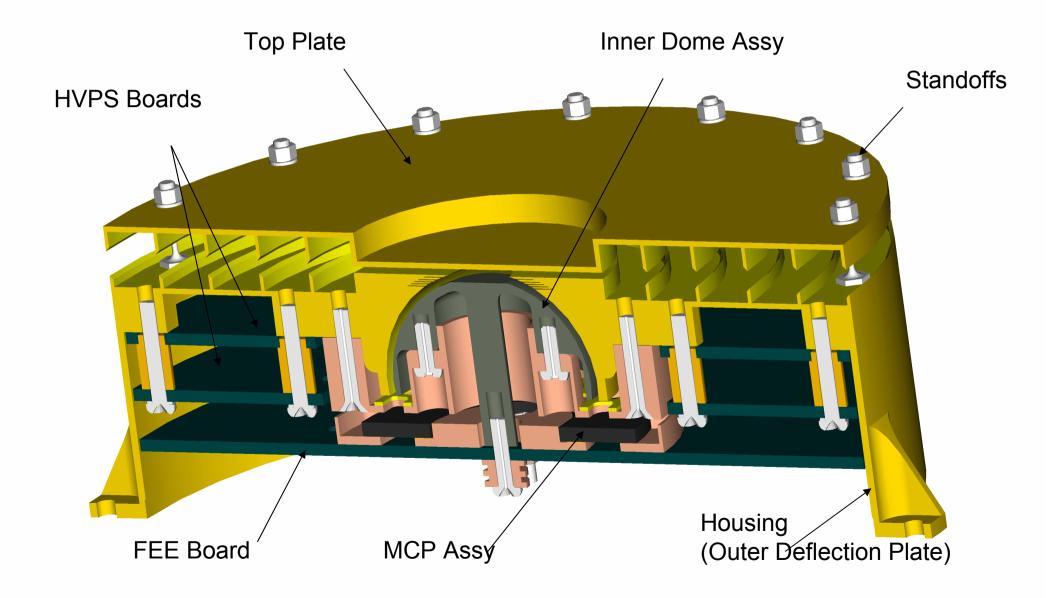
The Mars Express Electron Spectrometer

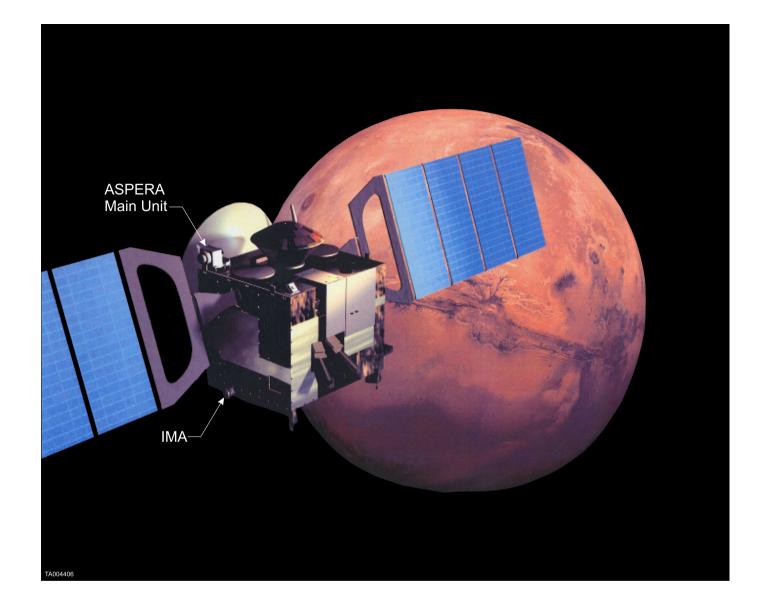


The Mars Express Analyzer of Space Plasmas and Energetic Atoms Experiment Main Unit



Cross-Section of the Mars Express Electron Spectrometer

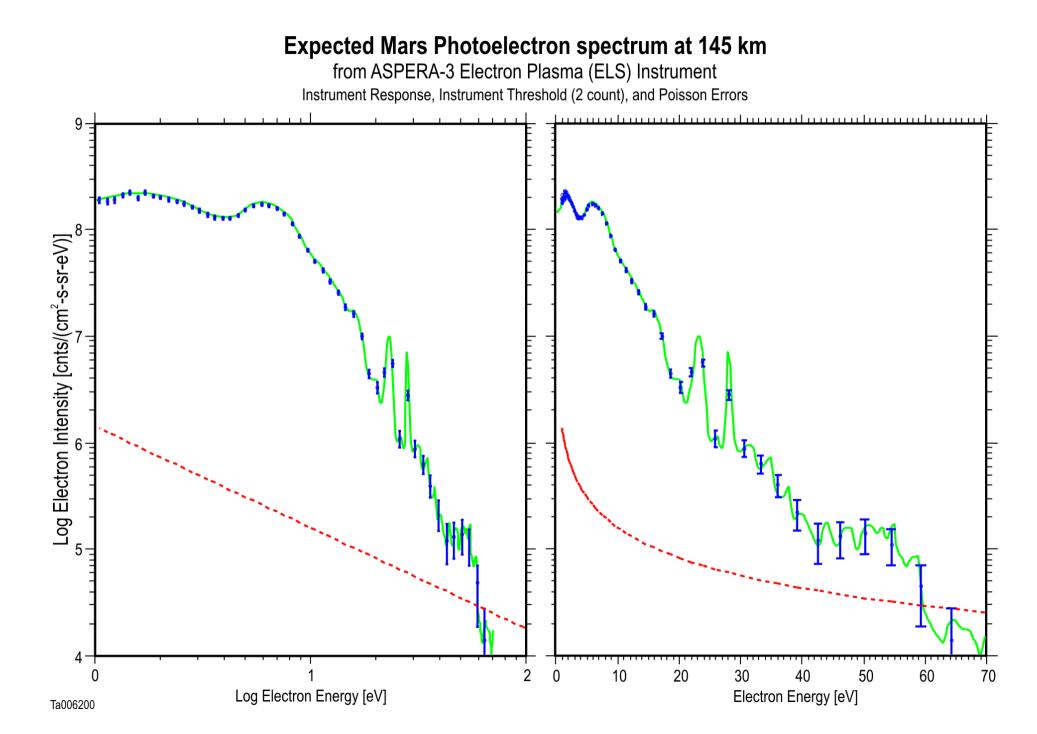




Mars Express was launched on June 2, 2003 and executed its orbital insertion burn on December 25, 2003.

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Expected Electron Spectrum at Mars



30.4 nm Photoionization of Carbon Dioxide

 $CO_{2} + hv \longrightarrow CO_{2}^{+} + e^{-} \begin{cases} X^{2}\Pi_{g} \text{ ionization potential } 13.8 \text{ eV} \longrightarrow 27.0 \text{ eV} \text{ electron} \\ A^{2}\Pi_{u} \text{ ionization potential } 17.7 \text{ eV} \longrightarrow 23.1 \text{ eV} \text{ electron} \\ B^{2}\Sigma_{u}^{+} \text{ ionization potential } 18.1 \text{ eV} \longrightarrow 22.7 \text{ eV} \text{ electron} \\ C^{2}\Sigma_{g}^{+} \text{ ionization potential } 19.4 \text{ eV} \longrightarrow 21.4 \text{ eV} \text{ electron} \end{cases}$

[Padial et al., 1981]

30.4 nm Photoionization of Atomic Oxygen

TA006169 AGU

 $O + h_{V} \longrightarrow O^{+} + e^{-} \begin{cases} {}^{4}S \text{ ionization potential } 13.62 \text{ eV} \longrightarrow 27.16 \text{ eV electron} \\ {}^{2}D \text{ ionization potential } 17.10 \text{ eV} \longrightarrow 23.68 \text{ eV electron} \\ {}^{2}P \text{ ionization potential } 18.50 \text{ eV} \longrightarrow 22.28 \text{ eV electron} \end{cases}$

[Mantas and Hanson, 1979]

Solar Spectrum Between 30 and 50 nm

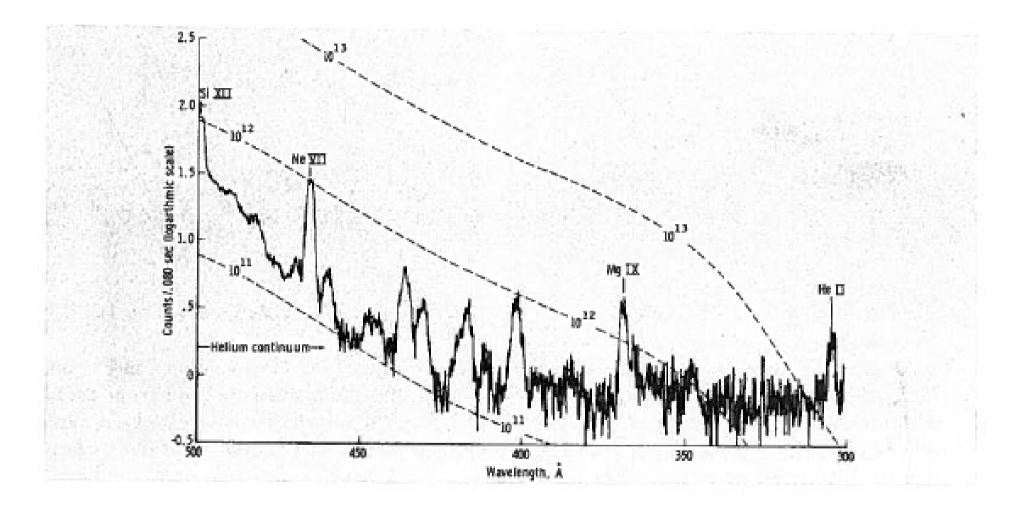
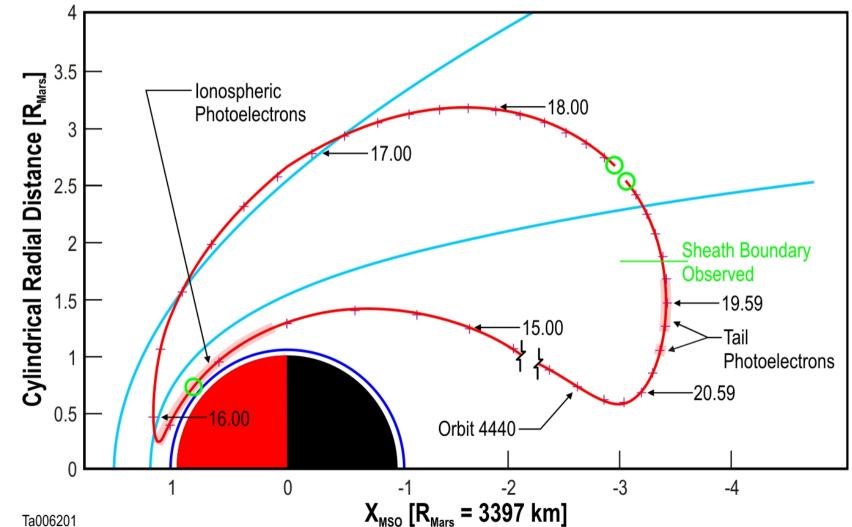


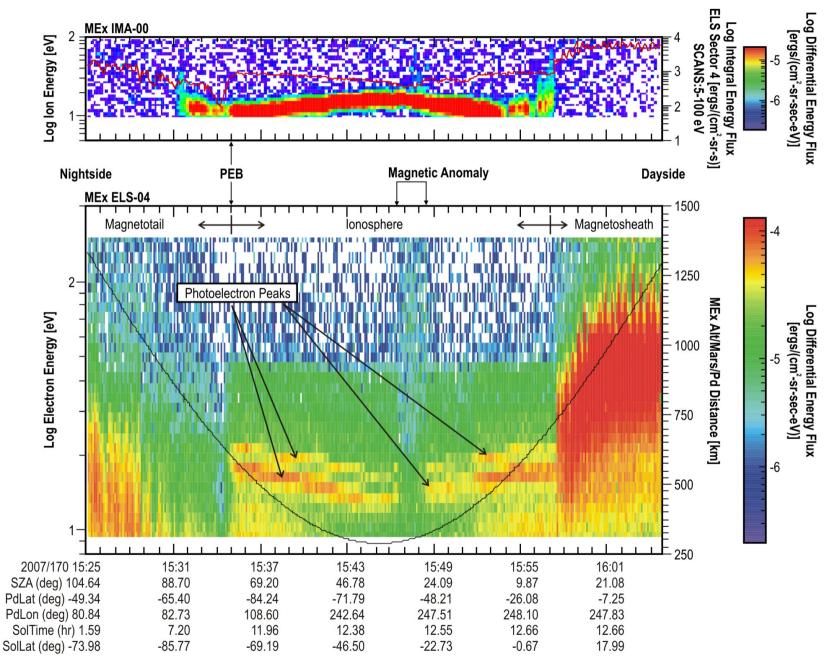
FIGURE 6-30 (concluded).—A detailed view of the spectrum from 1400 to 300 Å recorded by OSO 4. Gibson E. G., The Quiet Sun, NASA SP-303, 1973

Mars Express Sample Orbit

MEX position every 10 min. Cylindric coordinates. Orbit 4439 Pericenter at 2007-06-19 15:45:04 Start at 2007-06-19 12:23:30 End at 2007-06-19 19:06:42

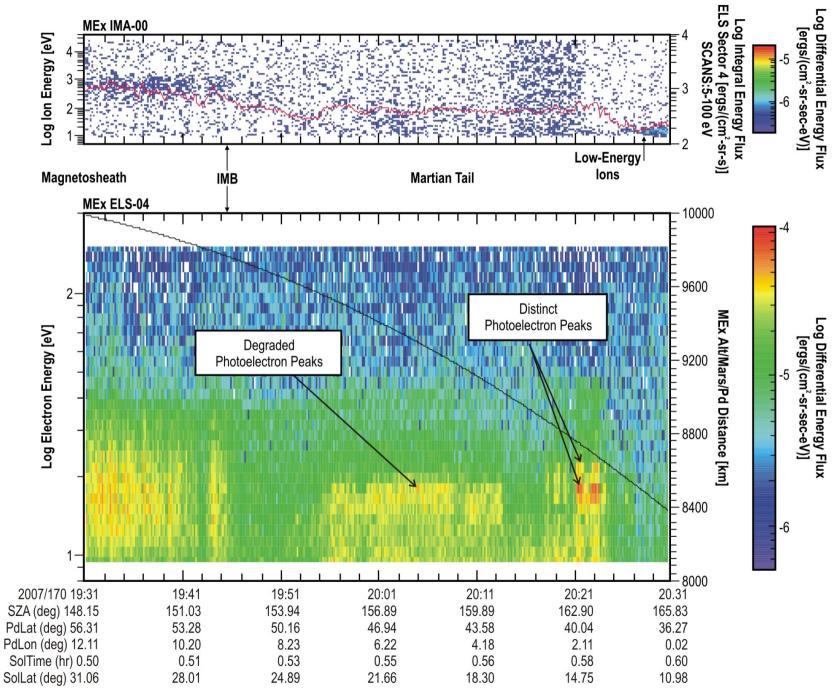


19 June 2007 Ionosphere Plasma



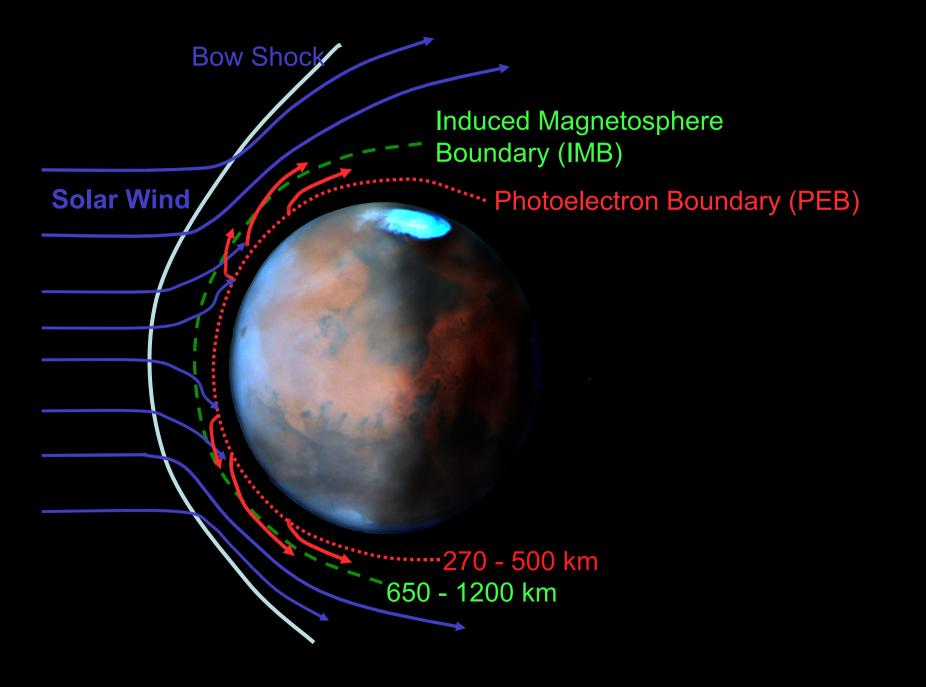
Ta006291

19 June 2007 Tail Plasma



Ta006204

Atmospheric Outflow from Mars

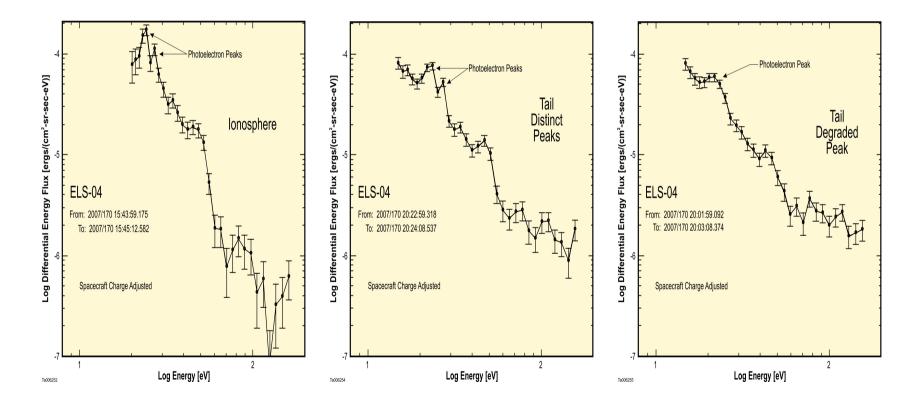


Ionosphere and Tail Electron Spectra

Α

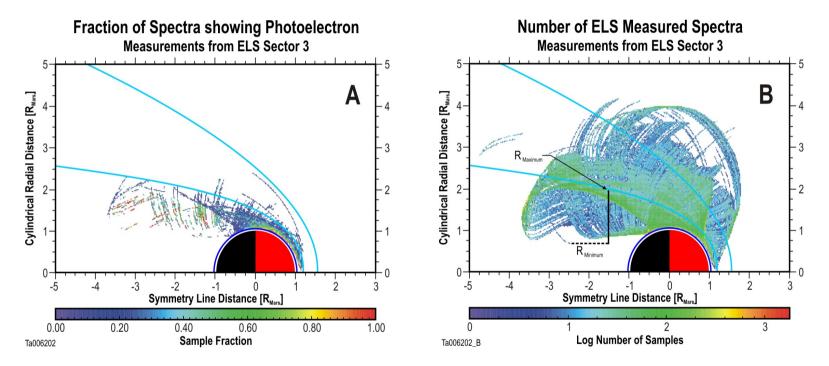
В





2004 Distinct Photoelectron Statistics

Jan 05, 2004 – Jan 25, 2005

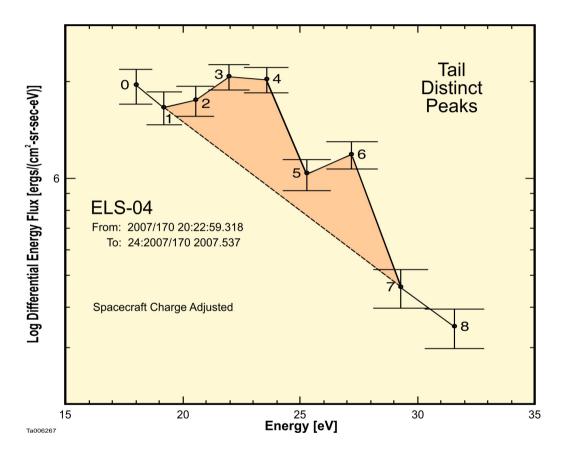


- 81,575 measured spectra less than 1.5 $R_{\mbox{\tiny Mars}}$ and below average MPB position
- 7,331 Photoelectron spectra less than 1.5 R_{Mars} and below average MPB position
- ELS measured distinct photoelectron spectra 9% of the time

Tail Area at 1.5 R_{Mars}

- $R_{max} = 6700 \text{ km}$ (average MPB position)
- $R_{min} = 2850 \text{ km}$ (min measured position)
- Annular Area = $1.16 \times 10^{18} \text{ cm}^2$

Electron Flux Integration Details



Escaping Electron: 5,74 x 10⁶ electrons/(cm² s sr) Uncertainty: 1.26 x 10²⁶ electrons/(cm² s sr)

Angular Measurement Range

Measurements typically between 1 and 3 ELS sectors.

Average angular area (2 ELS sectors): 0.478 ± 0.235 sr

Estimation of Electron Outflow from Mars in 2004

Electron Outflow

=	Electron Flux	*	Angular Flow	*	Escape Area	*	Yearly Measured Fraction
=	5.74 x 10 ⁶	*	0.478	*	1.16 x 10 ¹⁸	*	0.09

= $2.85 \pm 1.53 \times 10^{23}$ electrons/s

2004 Electron loss = 15 ± 8 Mmole

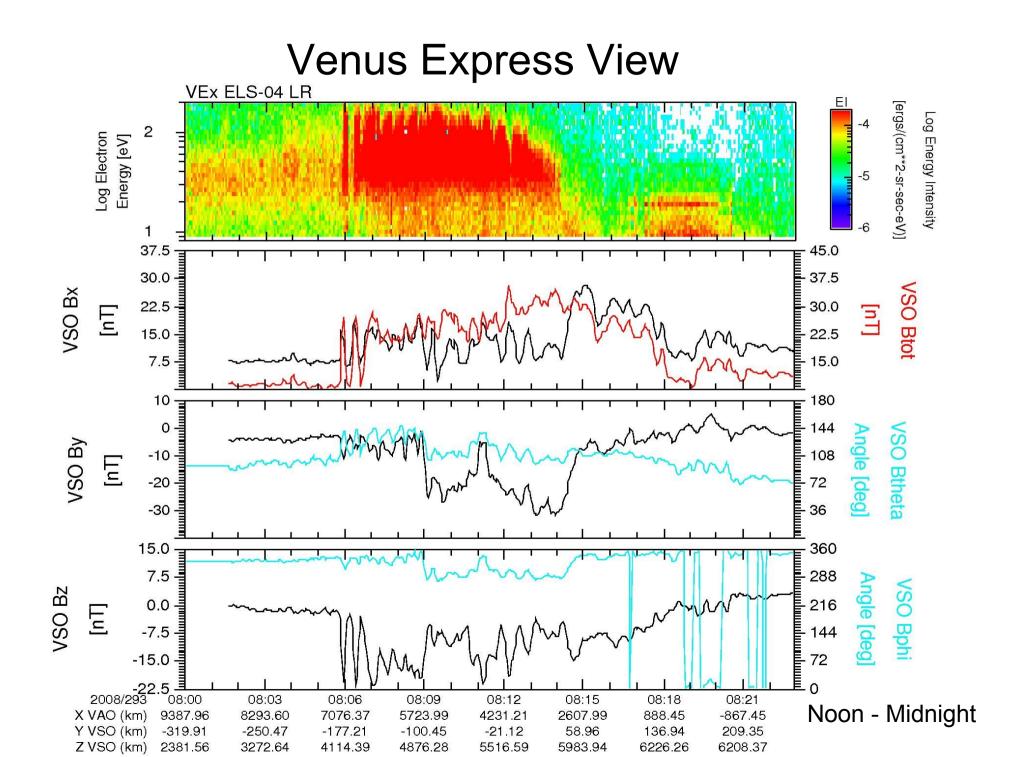
Acknowledgement: NASA Contract NASW-00003

How much is 15 Mmole?

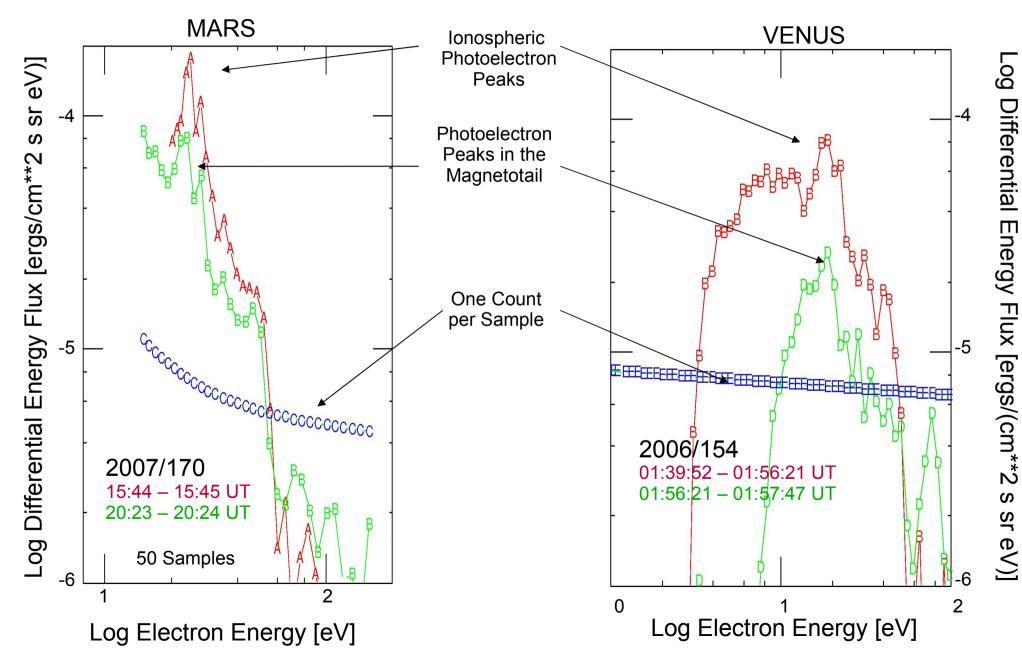
We assume that all the escaping electrons from Mars are due to carbon dioxide and that the escape rate is the same for all time.

Mars is approximately 5 billion years old.

In 5 billion years, if all of the escaping carbon dioxide formed a frozen polar cap 1 mm thick above 80° latitude, this is the quantity which would have been lost from the planet since it was formed.



Electron Escape Down the Obstacle Tail



Summary

During 2004, the average electron escape rate from Mars due to Solar Helium 30.4 nm radiation was $2.85 \pm 1.53 \times 10^{23}$ electrons/s.

For all of 2004, this electron outflow amounted to 15 ± 8 Mmole of electrons.

In 5 billion years, an equivalent amount of carbon dioxide amounting to a polar cap 1 mm thick above 80° latitude would have been lost under current escape rates.

Acknowledgments

This presentation acknowledges the generous support from the NASA under contract NASW-00003 at Southwest Research Institute in the United States of America, as well as the individual national agencies and institutions which support the Mars Express program in Sweden and the United Kingdom.

BACK-UP SLIDES

Comparison to Ion Outflow

Object – Determine the number of electrons created by ionization of solar 30.4 nm HeII which escape Mars.

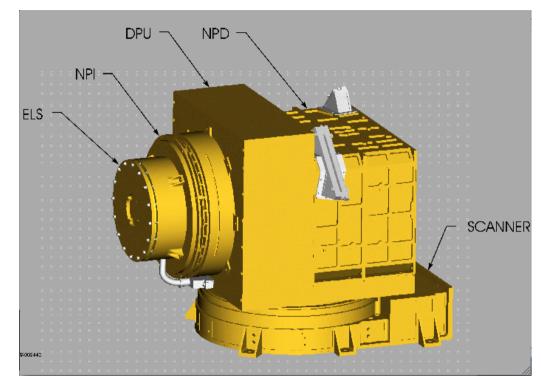
Ion Escape

- Total Phobos-2 at Solar Max: 3 x 10²⁵ s⁻¹ [Lundin et al., 1989,1990]
- Total MEX at Solar Min: 6-60 x 10²³ s⁻¹ [Dubinin et al., 2006]
- CO_2^+ scaled from Phobos-2: 4 x 10²⁴ s⁻¹ [Carlsson et al., 2006]
- O⁺ from MEX: 1.6 x 10²³ s⁻¹

 O_2^+ from MEX: 1.5 x 10²³ s⁻¹

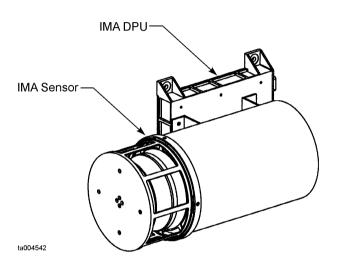
 CO_2^+ from MEX: 8 x 10²² s⁻¹ [Barabash et al., 2007]

ASPERA Instrumentation



Main Unit:

- Neutral particle imagers (NPI, NPD)
- Electron spectrometer (ELS)
- Data processing unit (DPU)
- Mechanical scanner



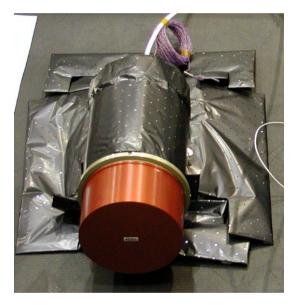
Ion Mass Analyzer (IMA)

Packaged for Pre-launch Testing



Main Unit:

- Neutral particle imagers (NPI, NPD)
- Electron spectrometer (ELS)
- Data processing unit
- Mechanical scanner



• Ion Mass Analyzer

Electron Spectrometer (ELS)

The Electron Spectrometer (ELS) sensor is a light-weight, low-power, spherical tophat electrostatic analyzer with collimation, detection, and readout system.

It measures the electron spectrum: electrons/cm² s sr eV vs. energy

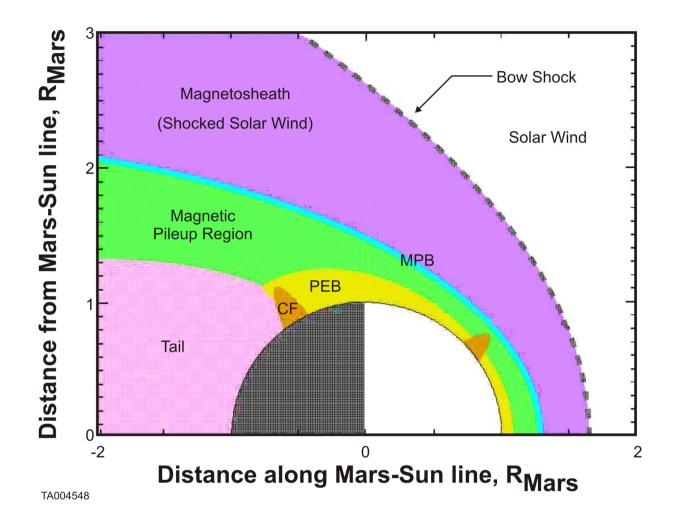
Energy range: 1 eV/q to 20 keV/q, one energy sweep (128 energy levels) per four seconds.

Energy resolution is 7%.

There are 16 anodes around a 360° fov, each defining a 22.5 ° sector.

Geometric factor (per sector) is 7.5 x 10 -5 cm² sr.

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One of the topics of study here on the Earth is the aurora borealis (northern lights).



AURORAL SHEETS



PICTURE TAKEN IN SAN ANTONIO IN 1989



A BRIGHT AURORAL DISPLAY



A STABLE ARC



Getting Ready for Launch in Kazakhstan



Soyuz - Fregot Launch Vehicle

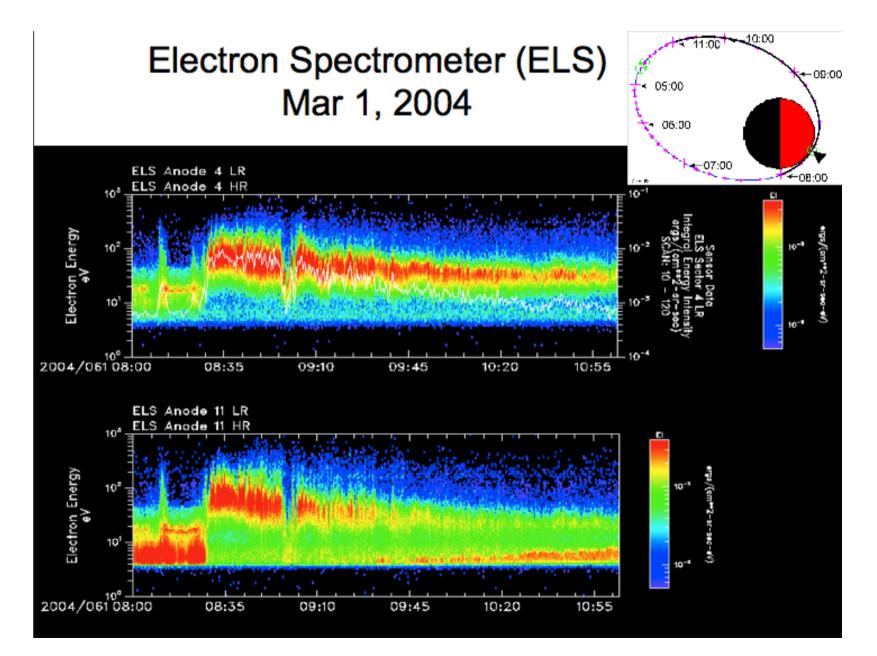


Launch Day June 2, 2003

Arrival on December 25, 2003



Real Data from Mars



Calibration

One of the most important things we do.

