

Simultaneous Photoelectron and Ion Measurements in the Martian Tail

R. A. Frahm, J. D. Winningham, J. R. Sharber (Southwest Research Institute, 6220 Culebra Road, San Antonio, Texas 78228, United States), R. Lundin, H. Nilsson, S. Barabash (Swedish Institute of Space Physics, Box 812, Kiruna S-981 28, Sweden) A. J. Coates, D.R. Linder (Mullard Space Science Laboratory, University College London, Holmbury St. Mary, Dorking Rh5 6NT, United Kingdom), A. Fedorov, and J. -A. Sauvaud (Centre d' Etude Spatiale des Tayonnements, 9 Avenue de Colonel Roche, Toulouse 31028, France)

ements of electrons by the Electron Spectrometer (FLS), ions by the lon Mass Analyzer (IMA), and neutral particles by the Neutral Particle Imager (NPI) and the Neutral Particle Detector (NPD). While orbiting Mars, the ELS is able to observe peaks in the photoelectron spectrum due to photoionization of carbon dioxide and atomic oxygen by Solar Helium 30.4 nm changes in the source of these peaks is the dayside Martia incosphere, with the majority of photoelectrons reach the spacecraft. By the time the atmospheric photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectrons reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectron is reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of lesse photoelectron is reach the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail of Mars, the case where the density is greatest. A fraction of the spacecraft in the tail o

Since December 25, 2003, the Mars Express (MEx) space (raft has been orbiting Mars. MEx measurements of the Martian particle environment have been performed by the Applyzer (IMA) and peutral particle masures electrons with the Neutral Particl

of photoelectron peaks in the energy spectrum are mainly due to ionization of carbon dioxide with Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be ionized by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be included by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be included by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be included by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen, which is also known to be included by Solar 30.4 nm photons and produce peaks in the electron speater than atomic oxygen atomic oxygen atomic oxygen atomic oxygen atom of carbon dioxide. The energy resolution of ELS (about 8%) is too large to distinguish between the energy spectrum of electrons photoionized from atomic oxygen. At altitudes above about 210 km, atomic oxygen becomes the dominate species over carbon dioxide electrons from ionization (at spacecraft altitudes) of the atomic oxygen. At altitudes above about 210 km, at

After the carbon dioxide or atomic oxygen are ionized, the charged components are subject to the local magnetic field and are transported accordingly [Mantas and Hanson, 1979]. The ASPERA-3 ELS has observed photoelectron peaks in the Martian induced magnetosphere while orbiting the planet at various altitudes [Frahm et al, 2006a,b]. During the nominal mission of Mars Express, the IMA sensitivity was tuned to measure escaping higher-energy ions and the capability to observe low-energy ions was limited. Late in the spring of 2007 the IMA was reprogrammed to increase its sensitivity to lower-energy ions. This resulted in the detection of low-energy planetary ions in the Martian ionosphere which were components of carbon dioxide. Ions in the Martian ionosphere which were components of carbon dioxide. Ions in the Martian ionosphere which were components of carbon dioxide. Ions in the Martian ionosphere which were components of carbon dioxide. Ions in the Martian ionosphere which were components are subject to the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express, the IMA sensitivity was tuned to make the capability of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field and are transported accordingly [Mantas and the capability to observe low-energy ions was limited. Express of the local magnetic field an

ectron Tail Observations

Many observations of photoelectron peaks in the tail of Mars have shown that the peaks are strong features which dominate the electron spectrum. Photoelectron peaks in the fall of Mals have shown that the peaks are strong leadness which dominate the electron spectrum. Photoelectron peaks in the fall are readily identifiable and consistent. Examples of photoelectrons in the fall of Mars are shown on 15 July 2007. Figure 1 shows the MEx spacecraft orbit around Mars in cylindrical coordinates with the location of the spacecraft marked every 10 minutes. Highlighted are locations where hotoelectron peaks are observed. During this transit, photoelectron peaks are observed by different sectors of the ELS <mark>it different times. Highlighted areas are marked with the ELS sector</mark> exhibiting the strongest photoelectron peak signal using the format ELS-xx, where xx is the sector number. ELS sectors measure 22.5° around a 360° entrance plane.

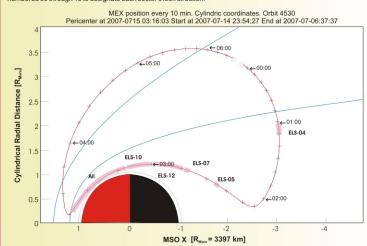


Figure 1. The Orbit of the Mars Express Spacecraft on 15 July 2007. The spacecraft orbit is shown in a cylindrical Marscentered Solar Orbital (MSO) coordinate system with the Sun at the left. Sketched in blue are the average locations of the bow shock, magnetic pile-up boundary, and photoelectron boundary as determined using Mars Global Surveyor data (Yignes et al., 2000). The location of the spacecraft is marked in UT every 10 minutes and labeled every hour. Locations where the photoelectron peaks are observed are highlighted. When the photoelectron peaks are prominent within an ELS ctor, this is noted near the corresponding highlighted region.

Figure 2 shows the energy-time spectrograms for each ELS sector between about 3000 km altitude and periapsis in units of differential energy flux, with the color scale adjusted to highlight the peaks in the photoelectrons. Fluxes in the dayside ionosphere are observed when the Solar Zenith Angle (SZA) is less than 90°. A comparison of the photoelectron peaks which are observed by various sectors illustrate that outside the ionosphere, the peaks only appear in selected sectors and not in all sectors as they do in the dayside ionosphere. For the case of this pass, the scanner is arked with a 10° offset, which places sectors 0, 13, 14, and 15 viewing toward the spacecraft body and sector 12

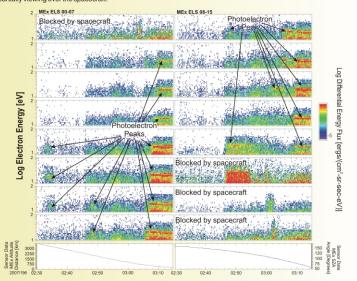


Figure 2. Energy-Time Spectrogram of Electrons Measured on 15 July 2007 from an Altitude of 3000 km to Periapsis Shown are spectrograms from the 16 ELS sectors (first 8 on the left, second 8 on the right, numbered lowest at top to highest at bottom) above drawings of the spacecraft altitude (left) and solar zenith angle (right). The color scale for the differential energy flux values has been adjusted to emphasize observations of photoelectron peaks.

In order to further illustrate the flux spectrum, we show four examples of the electron spectrum in Figure 3. In Figure 3a, we show the differential number flux of electrons from ELS-05 averaged between 02:32:32 UT and 02:34:46 UT (about 3000 km altitude in the tail) and 03:13:59 UT to 03:16:14 UT (near dayside periapsis). The overall spectrum at 3000 km is not as intense as the periapsis spectrum, which is a common feature observed; in general, the further away from the planet, the less intense the spectrum. However, in both cases, the photoelectron peaks are observed easily. It should be noted that the photoelectron peaks are shifted from their theoretical value and at periapsis (3000 km) the inferred spacecraft potential is about -4 volts (-8 volts).

The photoelectron peak is a strong feature at times, but it is not as strong at other times. Figure 3b illustrates the differential number flux of electrons from ELS-04 averaged between 01:03:30 UT and 01:08:06 UT (about 8900 km altitude in the tail) and 03:13:59 UT to 03:16:14 UT (near dayside periapsis). These photoelectron peaks are shifted from their theoretical value such that at 9800 km, the inferred spacecraft potential is about -14 volts. Due to the sampling of ELS, there are other times when the highest energy photoelectron peak is not prominent. Figure 3c illustrates the differential number flux of electrons from ELS-07 averaged between 02:47:28 UT and 02:49:43 UT (about 2100 km altitude in the tail) and 03:13:59 UT to 03:16:14 UT (near dayside periapsis). These photoelectron peaks are shifted from their theoretical value such that at 2100 km, the inferred spacecraft potential is about -5 volts both effects can combine at times to make identification difficult. In addition, the spacecraft can interact with ele in the peak, blurring their structure as in Figure 3d. Figure 3d illustrates the differential number flux of electrons from ELS-12 averaged between 02:50:29 UT and 02:52:44 UT (about 1800 km altitude in the tail) and 03:13:59 UT to 03:16:14 UT (near dayside periapsis). These photoelectron peaks are shifted from their theoretical value such that at 1800 km, the inferred spacecraft potential is about -1 volt. Here it should be noted that since the distinct peak structure

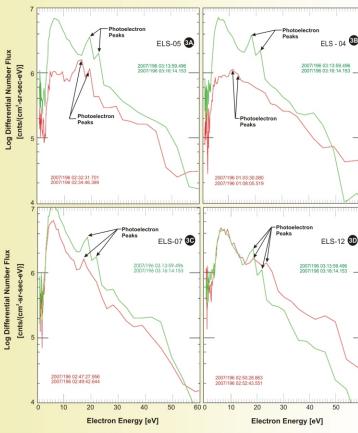


Figure 3. Selected Electron Energy Spectra. Four types of differential number flux spectra containing photoelectro peaks are presented. Each type is shown in red during tail measurements and green when measurements are in the dayside ionosphere. Spectrum a) occurs in ELS-05 and shows two peaks similar to the dayside ionosphere, b) occurs in ELS-04 and shows lower amplitude peaks in the tail, c) occurs in ELS-07 and shows only one photoelectron peak, and d)

Low-energy planetary ions are often observed in the Martian tail. For the same time period as the electrons shown by Figure 2, the ion energy-time spectrogram shown in Figure 4 is displayed in differential energy flux using a color range and energy scale appropriate to highlight the low-energy ions. Similar to the angular behavior of the electrons, the low-energy ion flux is highly directional, both in the tail and in the ionosphere. Ion fluxes between 0230 UT and 0256 UT are strongest in IMA-00, whereas in the ionosphere (after 0310 UT), fluxes are strongest in IMA-13. Between 0300 UT and 0307 UT, there are weak fluxes of low-energy ions observed in IMA-00.

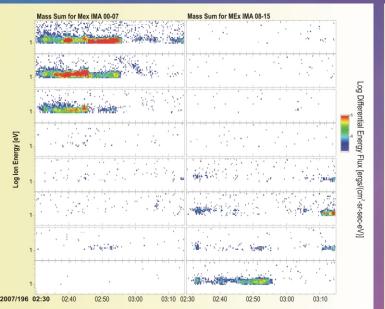


Figure 4. Energy-Time Spectrogram of lons Measured on 15 July 2007 from an Altitude of 3000 km to Periapsis. Shown are spectrograms from the 16 IMA sectors (first 8 on the left, second 8 on the right, numbered lowest at top to highest at pottom). The low-energy range detected by IMA and the color of the differential energy flux values has been adjusted to

The ion mass spectra corresponding to the selected electron spectra from Figure 3 are shown in Figure 5. In all cases, the vertical scale shows the ion energy between 5 eV and 50 eV, and the horizontal scale is expressed in terms of mass channel number. The mass channel number location of a species is dependent on the ion energy and instrument settings. For these measurements, the locations of where various mass species are recorded are indicated in each spectrum as overlaid lines with the species marked. All spectra are shown in units of differential energy flux. Figure 5a shows the join mass spectrum from IMA-00 averaged between 02:32:23 UT and 02:34:47 UT, and represents the ions detected in the range of 3000 km altitude in the Martian tail. Here the IMA shows masses in the region where ionization ordetected in the range of 3000 km alittude in the Martian Itali. Here the IMA shows masses in the region where roinzation products of carbon dioxide (including ionized atomic oxygen) are expected. Deeper in the tail (about 8900 km allitude) the IMA the ion mass spectrum from IMA-00 averaged between 01:03:23 UT and 01:08:11 UT (Figure 5b) resembles only noise and no detectable mass is observed. The effect of spacecraft charge of -14V should have accelerated low-energy ions into the energy range detectable by IMAI they were present; however, electron flux were lowest for this spectrum and it is very possible that the measurement threshold for flux in IMA is not enough to detections in this case, so this fact should be acknowledged requiring that fluxes at large tail distances be further examined.

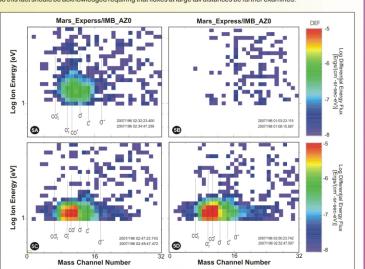


Figure 5. Selected Ion Energy Spectra. Four time periods of differential energy flux spectra containing low-energy for IMA-00 are presented. Locations of various species (ionization products of carbon dioxide) are drawn on top of ion mass channel spectra. Spectrum a) occurs during the time ELS-05 shows two photoelectron peaks, b) occurs during the time FLS-04 shows lower amplitude photoelectron peaks in the Martian tail, c) occurs during the time FLS-07 shows only one ron peak, and d) occurs during the time ELS-12 shows influence by the spacecraft.

02:49:47 UT. Here the IMA also shows ionization products of carbon dioxide (including ionized atomic oxagen) and the flux level at about 2100 km altitude is greater than that of the flux measurement at 3000 km by about an order of magnitude. Decreasing the altitude a bit continues to increase the flux of ions as shown in Figure 5d averaged from 02:50:24 UT to 02:52:48 UT in the range of 1900 km altitude. Again, Ionization products of carbon dioxide are

The increase of flux with decreasing altitude does not continue through to periapsis for this case. Since periapsis ion fluxes were greatest in IMA-13, the ion mass spectra for IMA-13 averaged from 03:13:60 UT to 03:16:24 UT are shown in Figure 6. The ion mass spectra shows that ionization products of carbon dioxide are observed and are in similar

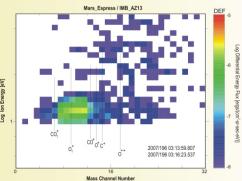


Figure 6. Perjapsis Ion Energy Spectra. Spectra shown in the jonosphere do not exhibit differential energy fluxes with as

ELS exhibits photoelectron peaks from the ionization of carbon dioxide and atomic oxygen, which in the Martian tail are directional. During times when these photoelectron peaks are observed, there exist low-energy ions. These low-energy ions are also directional in the Martian tail, but unlike the electrons which are mostly isotropic in the Martian ionosphere, nons the ionosphere are anisotropic (as shown here and in Frahm et al., 2007). On this pass, a rough correlation exists between the intensity of the differential energy flux of ions and the distance from the planet in the tail (Figure 5). Tail fluxes were shown to be more intense than those of the ionosphere. Frahm et al. [2007] showed an example of ion differential energy flux in the Martian ionosphere which was as intense as the fluxes shown in Figure 5d. During the measurements on 15 July 2007, IMA was not scanning in elevation. It is likely that the peak of the ion flux

Prior to the movement of the ASPERA-3 scanner in early 2006, the ELS and central IMA measurement planes were co-planar. However, the ASPERA-3 scanner is now used for minor scanning and positioning its electron and neutral imager sensors so that the Sun does not shine in the apertures of the neutral imaging instruments. This results in the asurement planes of ELS and IMA not being co-planar most of the time data is acquired. Thus, the IMA and ELS often do not view plasma flow in the same or a similar direction. During the data collection period presented here, the ELS measurement plane is tilted about 80° from the IMA measurement plane, with two intersection locations, the first being in the ELS-00 and ELS-15 sectors measuring about in the same direction as IMA-04 and IMA-05, and the second being in the ELS-07 and ELS-08 sectors measuring about in the same direction as IMA-12 and IMA-13. For reference, the ELS-04 and IMA-00 are viewing in the same direction if co-planer, but here they are separated by about 80°.

When surveying electron and ion spectra, times have been observed where there are low-energy ions but no photoelectron peaks and other times where there are photoelectron peaks but no low-energy ions. It is difficult to assess whether or not the low-energy ions are co-located with the photoelectrons which come from the dayside atmosphere of Mars because the ELS and IMA are not observing plasma in the same plane. Further data where ELS

These results presented here are (or may be) consistent with the escape mechanism suggested by Coates at al. [2007], that excaping photoeletrons at Titan set-up ambipolar electric field which pull ions out from the planetary ionosphere to form a polar wind. It is possible that a similar escape mechanism exists at Mars.

On a representative pass through the Martian tail region, in situ instruments of the ASPERA-3 experiment on Mars Express have measured low-energy ions concurrently with ionospheric photoelectron peaks. The photoelectron peaks can be observed as a single peak or multiple peaks due to the instrument response function of ELS and the energy of the peak as modified by the potential on the spacecraft. In cases where electrons from the photoelectron peaks encounter the spacecraft before detection, they can react with the spacecraft sheath and distort the electron

ideal and this leads to an ambiguity between the flow direction of low-energy ions and electrons in the photoelectron peaks. Under most circumstances, both the low-energy ions and electrons in the photoelectron peaks. as they travel down tail. Low-energy ions are comprised of cold planetary ions from the ionized components of carbon